

Aircraft Registration Profile

File ID: 2193Q Created: 03/27/2018 Changed: 03/27/2018

Company: N2193Q

Contact:

Address:

Alt Phone:

Fax:

Effective Date of Airworthiness Issued:

Aircraft Registration No: N2193Q

Comments:

Category: Airframe
Manufacturer: Cessna Aircraft Company
Model: 177RG

Position:
Part Number:
Serial Number: 177RG0593

File:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Model:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Event:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Remarks:	DOM 8-31-74			

Category: Engine
Manufacturer: Avco Lycoming
Model: IO-360-A1B6D

Position:
Part Number:
Serial Number: L-12393-51A

File:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Model:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Event:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Remarks:	OH 20 Dec 1991			

Category: Propeller
Manufacturer: McCauley
Model: B2D34C207

Position:
Part Number:
Serial Number: 742813

File:	Aircraft Hours: --	Date: --	TSN: --	CSN: --
Model:	Aircraft Hours: --	Date: --	TSN: --	CSN: --

nt: Aircraft Hours: -- Date: -- TSN: -- CSN: --
OH 1-30-12

ory: Magnelos
nufacturer: Bendix Corporation Position:
D-2000 SERIES Part Number: D4LN-201
Serial Number: 10-382555-11

: Aircraft Hours: -- Date: -- TSN: -- CSN: --
al: Aircraft Hours: -- Date: -- TSN: -- CSN: --
nt: Aircraft Hours: -- Date: -- TSN: -- CSN: --

ory: Air Filter
nufacturer: Brackett Aircraft Position:
BA-5710 Part Number:
Serial Number:

: Aircraft Hours: -- Date: -- TSN: -- CSN: --
al: Aircraft Hours: -- Date: -- TSN: -- CSN: --
nt: Aircraft Hours: -- Date: -- TSN: -- CSN: --
Has date code of 1994 on frame

ory: Ignition Systems Position:
nufacturer: Bendix Corporation Part Number: 10-357200-22
P/N 10-357XXX Serial Number:

: Aircraft Hours: -- Date: -- TSN: -- CSN: --
al: Aircraft Hours: -- Date: -- TSN: -- CSN: --
nt: Aircraft Hours: -- Date: -- TSN: -- CSN: --
Cessna P/N C292501-0105

Y: Fuel Injected System Position:
nufacturer: Precision Airmotive Part Number:
RSA-5AD1 Serial Number: 17670

: Aircraft Hours: -- Date: -- TSN: -- CSN: --
al: Aircraft Hours: -- Date: -- TSN: -- CSN: --
nt: Aircraft Hours: -- Date: -- TSN: -- CSN: --

FAA Airworthiness Directives Compliance Record

Aircraft Registration No: N2193Q

ATP Revision: 03/27/2018

Company:

Category: Airframe

Manufacturer: Cessna Aircraft Company

Model: 177RG

Position:

P/N:

S/N: 177RG0593

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2011-10-09 06/17/2011	[Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause,contd.	-- Hrs: -- C: --	PCW - Due every 100 hours or 12 months, whichever occurs first	Yes	D: -- Hrs: -- C: --	/
2008-26-10 C 01/05/2009	To prevent erroneous indications from the altimeter, airspeed, and vertical speed,contd.	-- Hrs: -- C: --	NA by date of replacement	No	D: -- Hrs: -- C: --	/
2000-06-01 05/05/2000	To prevent foreign material from entering the fuel system and engine, which could result in loss of engine power,contd.	-- Hrs: -- C: --	NA - not installed	No	D: -- Hrs: -- C: --	/
97-01-13 02/03/1997	TO PREVENT FUEL, OIL, OR HYDRAULIC SYSTEMS FAILURE CAUSED BY A COLLAPSED HOSE	-- Hrs: -- C: --	NA by date of hose replacement	No	D: -- Hrs: -- C: --	/
88-12-12 07/05/1988	TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE TO WATER CONTAMINATION OF THE FUEL SYSTEM	-- Hrs: -- C: --	PCW 12-18-88 tach 1514.4	No	D: -- Hrs: -- C: --	/
79-10-14 R1 05/30/1988	TO PROVIDE AN ALTERNATE SOURCE OF FUEL TANK VENTING IN CASE OF FUEL TANK VENT OBSTRUCTION BY FOREIGN MATERIAL, CONTD.	-- Hrs: -- C: --	NA by AC model	No	D: -- Hrs: -- C: --	/

Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Authorized By
86-24-07 01/07/1987	TO PREVENT ENGINE POWER INTERRUPTION DUE TO LOSS OF ATTACHMENT OF THE ENGINE CONTROLS	-- Hrs: -- C: --	PCW 4-24-87 tach 1339.9	No	D: -- Hrs: -- C: --	Signed
86-19-11 10/04/1986	TO ELIMINATE THE POSSIBILITY OF ENGINE POWER REDUCTION DUE TO CONTAMINATED FUEL	-- Hrs: -- C: --	PCW 11-28-86 tach 1329.4	No	D: -- Hrs: -- C: --	/
72-03-03 R3 10/15/1984	[Recurring] TO PREVENT INADVERTENT RETRACTION OF WING FLAP AND TO INSURE POSITIVE OPERATION OF THE ELECTRICAL WING, CONTD.	-- Hrs: -- C: --	NA by S/N and DOM	Yes	D: -- Hrs: -- C: --	/
79-08-03 C 06/06/1979	TO PREVENT ELECTRICAL SYSTEM FAILURE, SMOKE IN THE COCKPIT, AND/OR FIRE IN THE WIRE BUNDLE BEHIND THE INSTRUMENT PANEL	-- Hrs: -- C: --	PCW by SE78-48 10-02-78 tach 848.1	No	D: -- Hrs: -- C: --	/
77-12-08 06/27/1977	TO PREVENT UNWANTED PROPELLER ROTATION	-- Hrs: -- C: --	NA by S/N	No	D: -- Hrs: -- C: --	/
76-21-06 10/26/1976	TO PREVENT LOSS OF ENGINE OIL	-- Hrs: -- C: --	NA by S/N	No	D: -- Hrs: -- C: --	/
76-14-08 07/22/1976	TO ASSURE PROPER ATTACHMENT OF THE STABILATOR TRIM TAB ACTUATOR TO THE TRIM TAB LINKAGE	-- Hrs: -- C: --	NA by S/N	No	D: -- Hrs: -- C: --	/
76-04-03 02/26/1976	[Recurring] TO PRECLUDE RESTRICTIONS OF CONTROL MOVEMENT DUE TO JAMMING OF THE ARC PA-500A ACTUATOR GEAR TRAIN	-- Hrs: -- C: --	NA - Not installed	Yes	D: -- Hrs: -- C: --	/
75-07-02 03/26/1975	TO PRECLUDE SEPARATION OF THE FOAM RUBBER AIR FILTER SEAL	-- Hrs: -- C: --	PCW by SE75-3. See log book #1 entry 4-3-75 71.0 hours	No	D: -- Hrs: -- C: --	/

Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Let Authorized By Signed
74-16-06 08/09/1974	TO PREVENT FAILURE OR CHAFING OF THE OIL PRESSURE GAGE LINE LOCATED BETWEEN THE FIREWALL AND THE OIL PRESSURE GAGE	-- Hrs: -- C: --	NA by 5/74	No	D: -- Hrs: -- C: --	
68-17-04 09/07/1968	[Recurring] TO ASSURE PROPER OPERATION OF THE STALL WARNING SYSTEM IN FLIGHT	-- Hrs: -- C: --	NA by AC model	Yes	D: -- Hrs: -- C: --	

Category: Engine

Position:

ATP Revision: 03/27/2018

Manufacturer: Avco Lycoming

P/N:

Model: IO-360-A1B6D

S/N: L-12393-51A

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2017-16-11 08/15/2017	To prevent connecting rod failure which could result in uncontained engine failure, total engine power loss,contd.	-- Hrs: -- C: --	NA by date of eng OH and parts install	No	D: -- Hrs: -- C: --	
2015-19-07 11/03/2015	[Recurring] To prevent failure of the fuel injector fuel lines, which could lead to uncontrolled engine fire, engine,contd.	-- Hrs: -- C: --	PCW - due every 110 hours or fuel line work	Yes	D: -- Hrs: -- C: --	
2015-02-07 03/11/2015	To prevent the propeller governor shaft set screw from coming loose, causing damage to the engine and,contd.	-- Hrs: -- C: --	NA - has a rear mounted prop governor	No	D: -- Hrs: -- C: --	
2012-19-01 10/24/2012	To prevent failure of the crankshaft, which will result in total engine power loss, in-flight engine,contd.	-- Hrs: -- C: --	NA by engine S/N	No	D: -- Hrs: -- C: --	
2012-03-06 C 02/24/2012	To prevent an in-flight engine shutdown due to a failed fuel servo diaphragm, and damage to the airplane	-- Hrs: -- C: --	NA - Avstar fuel servo not installed	No	D: -- Hrs: -- C: --	

Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Authorized By Signed
2009-02-03 02/09/2009	[Recurring] To prevent a lean running engine, which could result in a substantial loss of engine power and subsequent,contd.	-- Hrs: -- C: --	NA by date of plug installation	Yes	D: -- Hrs: -- C: --	/
2006-06-16 04/27/2006	To prevent failure of the crankshaft, which could result in total engine power loss, in-flight engine,contd.	-- Hrs: -- C: --	NA by date of OH	No	D: -- Hrs: -- C: --	/
2005-19-11 10/21/2005	To prevent failure of the crankshaft, which could result in total engine power loss, in-flight failure, and,contd.	-- Hrs: -- C: --	NA by date of OH	No	D: -- Hrs: -- C: --	/
2004-10-14 C 06/25/2004	[Recurring] To prevent loosening or failure of the crankshaft gear retaining bolt, which may cause sudden engine failure	-- Hrs: -- C: --	Due after prop strike	Yes	D: -- Hrs: -- C: --	/
2003-14-03 08/14/2003	[Recurring] To prevent rotary fuel pump leaks, which could result in an engine failure, engine fire, and damage to or,contd.	-- Hrs: -- C: --	NA - Rotary pump not installed	Yes	D: -- Hrs: -- C: --	/
2002-12-07 07/03/2002	[Recurring] To prevent complete loss of engine oil and subsequent seizing of the engine and possibility of fire,contd.	-- Hrs: -- C: --	NA - converter plate not installed	Yes	D: -- Hrs: -- C: --	/
98-17-11 C 10/19/1998	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIGHT ENGINE FAILURE AND POSSIBLE,CONTD.	-- Hrs: -- C: --	NA by date of OH	No	D: -- Hrs: -- C: --	/
98-02-08 03/30/1998	[Recurring] TO PREVENT CRANKSHAFT FAILURE, WHICH CAN RESULT IN ENGINE FAILURE, PROPELLER SEPARATION, FORCED LANDING, AND,CONTD.	-- Hrs: -- C: --	NA - Constant speed prop installed	Yes	D: -- Hrs: -- C: --	/
97-15-11 08/12/1997	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE	-- Hrs: -- C: --	NA by date of cyl replacement	No	D: -- Hrs: -- C: --	/

Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
96-23-03 12/17/1996	TO PREVENT AN INFLIGHT ENGINE FAILURE DUE TO FUEL STARVATION, WHICH COULD RESULT IN A FORCED LANDING	Hrs: -- C: --	NA by P/N		D: -- Hrs: -- C: --	
96-09-10 C 07/15/1996	TO PREVENT OIL PUMP FAILURE DUE TO IMPELLER FAILURE, WHICH COULD RESULT IN AN ENGINE FAILURE	-- Hrs: -- C: --	PCW at engine OH	No	D: -- Hrs: -- C: --	
95-26-02 01/24/1996	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IN SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	-- Hrs: -- C: --	NA by model and N #	No	D: -- Hrs: -- C: --	
95-07-01 04/12/1995	TO PREVENT ENGINE FAILURE DUE TO CONNECTING ROD BOLT FAILURE WHICH COULD RESULT IN DAMAGE TO OR LOSS, CONTD.	-- Hrs: -- C: --	NA by date of OH	No	D: -- Hrs: -- C: --	
93-14-15 08/23/1993	TO PREVENT POSSIBLE PREMATURE ENGINE FAILURE	-- Hrs: -- C: --	NA - not modified by STC	No	D: -- Hrs: -- C: --	
92-12-05 07/10/1992	TO PREVENT PISTON PIN FAILURE, OR PISTON RELEASE, AND ENGINE FAILURE	-- Hrs: -- C: --	NA by date of OH	No	D: -- Hrs: -- C: --	

Category: Propeller

Position:

ATP Revision: 03/27/2018

Manufacturer: McCauley

P/N:

Model: B2D34C207

S/N: 742813

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
--		Hrs: -- C: --			D: -- Hrs: -- C: --	

Category: Magnetos

Position:

ATP Revision: 03/27/2018

Manufacturer: Bendix Corporation

P/N: D4LN-201

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	-- Hrs: -- C: --	NA by engine model installed on	Yes	D: -- Hrs: -- C: --	/
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	-- Hrs: -- C: --	PCW at OH 08-14-91	No	D: -- Hrs: -- C: --	/
82-11-05 06/09/1982	[Recurring] ENGINE POWER LOSS AND ENGINE DAMAGE RESULTING FROM FROM LOOSENESS OF THE DISTRIBUTOR GEAR ELECTRODE	-- Hrs: -- C: --	NA by S/N	Yes	D: -- Hrs: -- C: --	/
79-12-07 06/19/1979	[Recurring] TO DETECT LOOSE DISTRIBUTOR BLOCK BUSHINGS	-- Hrs: -- C: --	NA by model	Yes	D: -- Hrs: -- C: --	/

Category: Air Filter

Position:

ATP Revision: 03/27/2018

Manufacturer: Brackett Aircraft

P/N:

Model: BA-5710

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
81-15-03 07/20/1981	TO PREVENT POSSIBLE FAILURE OF THE ALUMINUM AIR FILTER RETAINER SCREEN OR GASKETS WITH POTENTIAL, CONTD.	-- Hrs: -- C: --	PCW - AC has latest version of air filter assy installed with Sep 1994 date stamped on assy	No	D: -- Hrs: -- C: --	/

Category: Ignition Systems

Position:

ATP Revision: 03/27/2018

Manufacturer: Bendix Corporation

P/N: 10-357200-22

Model: P/N 10-357XXX

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed

Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Signed
76-07-12 R1 08/30/1977	[Recurring] IGNITION SWITCHES	-- Hrs: -- C: --	PCW - Due every 100 hours	Yes	D: -- Hrs: -- C: --	/

ATP Revision: 03/27/2018

Category: Fuel Injected System

Position:

Manufacturer: Precision Airmotive

P/N:

Model: RSA-5AD1

S/N: 17670

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2009-02-03 02/09/2009	[Recurring] To prevent a lean running engine, which could result in a substantial loss of engine power and subsequent,contd.	-- Hrs: -- C: --	NA by date of servo plug install	Yes	D: -- Hrs: -- C: --	/

03/27/2018 8:06 PM

10/10